## IBEX Command Approval Checklist Rev 16h reduces battery cell balancing offset from maneuvers

Orbit	592	Special Ops	y con sanamemy enece			
14 R <sub>E</sub> asc. Date/Time	12/23/2022 0	5:06:44	15 R <sub>E</sub> asc. Date/Time	12/23/2022 06:57:00		
Apogee	12/27/2022 06:22:46	Maneuver Window Start	12/26/2022 12:23:48	Maneuver Window End	12/27/2022 16:23:50	
Apogee Target	targetX:0.10958000000000000 targetY:-0.90830999999999999999999999999999999999					
15 R <sub>E</sub> desc. Date/Time	12/31/2022 0	5:56:44	14 R <sub>E</sub> desc. Date/Time	12/31/2022 07:47:00		
Perigee	12/31/2022 17:17:14	Maneuver Window Start	12/31/2022 08:17:40	Maneuver Window Stop	1/1/2023 02:48:54	
Perigee Target	targetX:0.18836500000000000 targetY:-0.897035999999999994  targetZ:-0.3998050000000002					
Eclipse	No	Eclipse Start		Eclipse End		
Sun Mnvr	No	Apogee/Perigee		Sun Angle at DESCENDING		
Approved Version	IBEX_2022_356	_o0592a_v001.s	scr			

Activity	Command Checks	Date Done	Done By
Supporting Materials	<ol> <li>IBEX_CrossingTimes_<date>_v00x.txt on SFTP at /Archive-Incoming/IBEXIncoming/FDG/PredictedEphemeris/Orbit Events/.</date></li> <li>Orbit Events File on SFTP at /IBEXOutgoing/MOC/Moc-Soc/oef/.</li> <li>Command Constraint Violations Report on SFTP at /IBEXOutgoing/MOC/Moc-Soc/cvr/.</li> <li>Contacts this orbit Orbit_oXXX.txt included in the ATS approval email.</li> <li>Science Tasking File at /Archive-Incoming/IBEXIncoming/MOC/Soc-Moc/stf/.</li> <li>Merged ATS at <a href="http://ibex.unh.edu/cgi-bin/ats.cgi">http://ibex.unh.edu/cgi-bin/ats.cgi</a>.</li> </ol>	The Merged ATS does not include orbit 192 yet	NGA
Sun Maneuvers	<ul> <li>Additional contacts should not be planned to support IBEX Sun Precession Maneuvers due to star tracker outages. The standard apogee and perigee contacts should be used to verify that a maneuver has occurred. If it is not possible to plan one of the standard contacts after the star tracker outage is down to 50% and a valid quaternion reading can be made to verify the maneuver, the coarse Sun sensor angle and the thruster pulse count will be used to determine a) whether a maneuver took place, and b) whether the pointing after the maneuver is as expected +/- 2 degrees.</li> </ul>	N/A	
	<ul> <li>The nominal off-Sun pointing constraint is 7.25 degrees. Based on the missed maneuver in orbit 114, the payload team has determined that there is no hardware risk associated with off-Sun pointing up to angles of at least 13 degrees. There is a higher background noted in the data starting at around 9.5 degrees off Sun pointing.</li> </ul>		
File Input Check	<ol> <li>Current OEF inputs are Forecast STF, last orbit's OEF &amp; latest predictive ephemeris.</li> <li>ATS inputs are this orbit's OEF &amp; STF. (And ABS if present.)</li> <li>ATS filename is of the format IBEX_yyyy_doy_o0xxxa_v0zz.scr.         where IBEX is capitalized, yyyy is the year, doy is the day of year, xxx is the 3-digit orbit number and zz is the 2-digit version number.         Any special operations ATSs will have another designation between the orbit number and version number (i.e. *o0186a_hgc_v001 for the Hi gain curve).</li> </ol>	12/11/22	NGA

Eclipses	1.	Check OEF for eclipses during the orbit.	N/A	
Lonpood	2.	Verify long eclipse flag start & stop times reflect Ryan Tyler's recommendations based on his	14/7	
		eclipse diagnostic tool. Suggestions made by Ryan after the use of this tool trump the general		
		guidelines below. (Please note, specific timing may shift if the recommendations are relative to		
		eclipse timing. For example, it may say set LE flag to false X hours after the end of the eclipse		
		with a given FALSE time suggested. If the eclipse timing changes as the ephemeris becomes more refined, this command time may shift.)		
	3.	Verify no contacts planned during an eclipse. Note: If in conflict, the eclipse diagnostic		
	0.	recommendations will trump the general guidelines below.		
		a Verify transmitter OFF from 30 minutes before eclipse start through the end of the		
		eclipse.		
		b For an eclipse where the long eclipse flag is set, schedule a SOH contact directly		
		following the end of the eclipse (or per Ryan's assessment).		
	4.	c Set the LE flag according to Ryan's assessments.  Verify no maneuver planned during an eclipse. Note: If in conflict, the eclipse diagnostic		
	4.	recommendations will trump the general guidelines below.		
		a Verify no maneuver or cat bed heaters on from 3 hours before eclipse start through		
		3 hours after eclipse end.		
		b Verify no maneuver or cat bed heaters are on while the long eclipse flag is set.		
	5.	Verify the following additional constraints (from battery balancing section).		
		a Verify the first command sets the long eclipse flag to TRUE, the second command		
		sets the flag to FALSE.		
		b Verify P/L is in HVSTANDBY or HVENG.		
		c Verify no charging cycle within 2 hours of ASCENDING or DESCENDING macro		
		execution.		
	1.	This applies to all eclipses, not just moderate or long eclipses.  Verify STX on/off times, downlink rate against <i>Orbit oXXX.txt</i> file.		
Contact	2.	Verify contacts in the previous ATS have not been duplicated.	12/11/22	NGA
Commands	3.	Verify all currently planned contacts in Orbit_xxx.txt are in the ATS.		
	4.	Verify each contact contains the following 5 commands.		
		SetRelay stx,on		
		• SetDownlink2K (2K, 40K, 64K, 160K, or 320K)		
		SetBilevelOutputControlReg STXMODE_Strobe,ON		
		SetBilevelOutputControlReg COHERENT,ON		
		SetRelay stx,off		
	5.	If contact is near an eclipse		
		a. Verify transmitter OFF from 30 minutes before eclipse start through 30 minutes after eclipse		
		end.		
		b. If additional transmitter constraints exist, they will be captured in Ryan's recommendations.		
	6.	If an APL contact is being used for an SSR Dump, the data rate should be at least 160 ksps & the SSR DUMP NEW command should be included in the contact commands. Commands in Orange should only		
		be sent if the SSR pointers need to be reset this perigee. These commands should be separated by 2s and		
		occur 2s after the SSR_DUMP_NEW command.		
		SetRelay stx,on		
		SetDownlink2K		
		SetBilevelOutputControlReg STXMODE_Strobe,ON		
		SetBilevelOutputControlReg COHERENT,ON		
		SetDownlink320K		
		• SSR_DUMP_NEW		
		• SSR_SET_RD_PTR 70000		
1	1			
		<ul><li>SSR_SET_WRT_PTR 70000</li><li>SetRelay stx,off</li></ul>		

SC State	1.	Transition to Science state will be first command of each ATS (at 14 Re).	12/11/22	NGA
Science:		• SetScState science \$TIME=2022/12:23:05:06:59		
arc a	2.	Lo science mode will be the next command (at 14 Re).		
		LO_SCIENCE_MODE NORMAL		
	3.	Verify no transition to Science again at the end of the ATS. The ATS commands go from 14 Re to		
		14 Re in each orbit.		
	4.	Verify the transition to Science commands for this orbit are not part of the previous ATS using		
		http://ibex.unh.edu/cgi-bin/ats.cgi.		
	5.	Verify that the beginning of this ATS does not overlap with the end of the previous orbit's ATS		
		using http://ibex.unh.edu/cgi-bin/ats.cgi.		
Payload	1.	Verify w/ Crossing Times report that it occurs about 15 Re ascending. The arc a ASCENDING	12/11/22	NGA
Mode		commands can start any time at or above 15 Re ascending.	12/11/22	110/1
HVSCI:	2.	Verify commanding takes ~ 24 minutes.		
arc a		• ASCENDING_PL1 \$TIME=2022/12:23:06:57:12		
		ASCENDING_HI		
		SET_PARAMETER 1, TLM_RATE_SOH		
		SET_PARAMETER 4, HV_STEP_DWELL		
		SET_PARAMETER 3, HV_STEP_FRAC		
		HI_COL_NEG_LVL 1400		
		• CEU_HI_CEM_1_LVL 1780		
		• CEU_HI_CEM_2_LVL 1780		
		• CEU_HI_CEM_3_LVL 1780		
		• CEU_HI_CEM_4_LVL 1900		
		SET_PARAMETER 0, TLM_RATE_SOH		
		ASCENDING_PL2		
		ASCENDING_PL1		
		ASCENDING LO		
		SET PARAMETER 1, TLM RATE SOH LO		
		SET PARAMETER 4, HV STEP DWELL		
		SET PARAMETER 1, HV STEP FRAC		
		CEU LO IF MCP VSET 2901		
		SET_PARAMETER 0, TLM_RATE_SOH_LO		
		• ASCENDING PL2 \$TIME=2022/12:23:07:21:31		
Payload	1.	Payload DESCENDING commands end 1.5h before thruster enable.	12/11/22	NGA
Mode	2.	Verify commands take ~23 minutes to execute.		
HVSTANDBY		• DESCENDING_PL1 \$TIME=2022/12:26:10:27:50		
: arc a		DESCENDING_LO     ASSENDING_RIG		
		ASCENDING_PL2		
		• DESCENDING_PL1		
		DESCENDING_HI		
		• DESCENDING_PL2 \$TIME=2022/12:26:10:50:40		
00.01.1	1.	Spacecraft Housekeeping command occurs 1h before thruster enable.	40/44/00	NGA
SC State		• SetScState housekeeping \$TIME=2022/12:26:11:23:50	12/11/22	NGA
HK : arc a		- Seissitate Housekeeping 3 Hivit - 2022/12.20.11.23.30		

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Inertial	1.	Use this command sequence if an apogee inertial maneuver is used, otherwise skip to the 'Sun	12/11/22	NGA
Maneuver:	_	Precession Maneuver: Apogee' sequence below.		
Apogee	2.	Verify Thruster enable command occurs within STF maneuver window.		
	3.	Verify no eclipse occurs from cat bed heater on through set FC mode Mission.  Verify cat bed heaters come on 55 min before burn.		
	4.	·		
	_	CATBED_5N_HTR,ON \$TIME=2022/12:26:11:28:48  Voils   Volume   Filter in put a plant in ground appropriate   8 paties step up data in disable d		
	5.	Verify Kalman Filter input select is ground command & estimator update is disabled.		
		• SetKFInputSelect GND_CMD, 0, 0, 0, 0		
		SetEstUpdateEnables ENABLE, DISABLE		
	6.	Verify in FC mode Burn.		
		SetFcMode burn		
	7.	Compare SetInrDir in ATS with pointing as defined in the Forecast STF & verify the vectors match.		
		• 0.10958,-0.90831,-0.403689		
		targetX:0.10958000000000000 targetY:-0.90830999999999999999999999999999999999		
		targetZ:-0.403689000000002		
	8.	Verify inertial maneuver chosen.		
	0.	SetLrTarget ACS_INERTIAL		
	9.	Verify thrust time set to 11 min.		
	9.	SetThrustTime 660		
	10			
	10.	Verify thruster enable command matches RepointingManeuverStart time in OEF.		
		SetThrustEnable ENABLE \$TIME=2022/12:26:12:23:48  Paralistic PM-results (2022 42:26:12:23:48)  Paralistic PM-results (2022 42:26:12:23:48)		
		RepointingManeuverStart 2022-12-26T12:23:48  Verify 10 print to a first three transport of the part of the pa		
	11.	Verify 10 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set to 0.		
		SetThrustEnable DISABLE \$TIME=2022/12:26:12:33:48		
		SetHTRCmd CATBED_5N_HTR,OFF		
	42	SetThrustTime 0  Verify 25 min of the thrustone and blad. Static 7 materials automa (Verify act. FC made Mission).		
	12.	Verify 25 min after thrusters enabled: Static Z rate set, outage %valid set, FC mode Mission.		
		• SetStaticZrate ESTIMATOR, 0.418 \$TIME=2022/12:26:12:48:58		
		SetKFInputSelect STA_PCT_VALID, 43,28,33,48		
	-	SetFcMode Mission		
Sun	1.	Use this command sequence in the event of an apogee Sun maneuver.	N/A	
Precession	2.	Verify Thruster enable command occurs within STF maneuver window.		
Maneuver	3.	Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
: Apogee	4.	Verify cat bed heaters powered on 55 min before thruster enable.		
	_	• CATBED_5N_HTR,ON		
	5.	Verify in FC mode Burn and Sun target.		
		SetFcMode burn		
		SetLrTarget ACS_SUN		
	6.	Verify thrust time set to 16 min.	1	
	_	SetThrustTime 960		
	7.	Verify thruster enable command matches SunMvrBegin time in OEF.		
		SetThrustEnable ENABLE		
		SunMvrBegin		
	8.	Verify 15 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
		to 0, FC mode Mission.		
		SetThrustEnable DISABLE	1	
		SetHTRCmd CATBED_5N_HTR,OFF		
		SetThrustTime 0		
		SetFcMode Mission		
SC State	1.	Spacecraft Science commands occur ~1h after thruster Disable.	12/11/22	NGA
Science :		• SetScState science \$TIME=2022/12:26:13:38:54	12/11/22	NGA
arc b		LO_SCIENCE_MODE NORMAL	1	
aicu	1		L	

## IBEX Command Approval Checklist K. Fairchild

Rev 17 Last update 11/05/2022

Payload Mode HVSCI: arc b	1. 2.	Verify payload ASCENDING commands begin at least 1.5 hours after thruster DISABLE.  Verify commands take ~24 minutes to execute.  ASCENDING_PL1 \$TIME=2022/12:26:14:08:54  ASCENDING_HI  SET_PARAMETER 1, TLM_RATE_SOH  SET_PARAMETER 4, HV_STEP_DWELL  SET_PARAMETER 3, HV_STEP_FRAC  HI_COL_NEG_LVL 1400  CEU_HI_CEM_1_LVL 1780  CEU_HI_CEM_2_LVL 1780  CEU_HI_CEM_3_LVL 1780  CEU_HI_CEM_4_LVL 1900  SET_PARAMETER 0, TLM_RATE_SOH  ASCENDING_PL2  ASCENDING_PL1  ASCENDING_LO  SET_PARAMETER 1, TLM_RATE_SOH_LO  SET_PARAMETER 4, HV_STEP_DWELL  SET_PARAMETER 1, HV_STEP_DWELL  SET_PARAMETER 1, HV_STEP_FRAC  CEU_LO_IF_MCP_VSET 2901  SET_PARAMETER 0, TLM_RATE_SOH_LO  ASCENDING_PL2 \$TIME=2022/12:26:14:33:13	12/11/22	NGA
Payload Mode HVSTANDBY : arc b	1.	Verify w/ Crossing Times report that it occurs about 15 Re descending. The arc b DESCENDING commands can complete any time at or above 15 Re descending.  Verify commands take ~23 minutes to execute.  • DESCENDING_PL1 \$TIME=2022/12:31:05:30:43  • DESCENDING_LO  • ASCENDING_PL2  • DESCENDING_PL1  • DESCENDING_PL1  • DESCENDING_PL2 \$TIME=2022/12:31:05:53:33	12/11/22	NGA
SC State HK : arc b	1.	Verify with Crossing Times report that Transition to Housekeeping state occurs at 14 Re desc (or an hour before the maneuver if the maneuver occurs less than 1 hour after 14 Re desc).  • SetScState housekeeping \$TIME=2022/12:31:07:17:01	12/11/22	NGA

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Inertial	1.	Use this command sequence if a perigee inertial maneuver is used, otherwise skip to the 'Sun	12/11/22	NGA
Maneuver:	_	Precession Maneuver : Perigee' sequence below.		
Perigee	2.	Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
	3.	Verify cat bed heaters come on 55 min before burn.		
		• CATBED_5N_HTR,ON \$TIME=2022/12:31:07:22:40		
	4.	Verify in Housekeeping state.		
	5.	Verify Kalman Filter input select is ground command & estimator update is disabled.		
		<ul><li>SetKFInputSelect GND_CMD, 0, 0, 0, 0</li></ul>		
		SetEstUpdateEnables ENABLE, DISABLE		
	6.	Verify in FC mode Burn.		
		SetFcMode burn		
	7.	Compare SetInrDir in ATS with target vector in the Forecast STF & verify the vectors match.		
		• SetInrDir 0.188365,-0.897036,-0.399805		
		targetX:0.18836500000000000 targetY:-0.8970359999999994		
		targetZ:-0.3998050000000002		
		Verify inertial maneuver chosen.		
		SetLrTarget ACS_INERTIAL		
	8.	Verify thrust time set to 11 min.		
		SetThrustTime 660		
	9.	Verify thruster enable command matches RepointingManeuverStart time in OEF.		
		SetThrustEnable ENABLE \$TIME=2022/12:31:08:17:40		
		RepointingManeuverStart 2022-12-31T08:17:40		
	10.	Verify 10 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
		to 0.		
		<ul> <li>SetThrustEnable DISABLE \$TIME=2022/12:31:08:27:40</li> </ul>		
		• CATBED_5N_HTR,OFF		
		SetThrustTime 0		
	11.	Verify 25 min after thrusters enabled: Static Z rate set, outage %valid set, FC mode Mission.		
		SetStaticZrate ESTIMATOR, 0.418 \$TIME=2022/12:31:08:42:50		
		<ul> <li>SetKFInputSelect STA_PCT_VALID, 43,28,33,48</li> </ul>		
		SetFcMode Mission		
Sun	1.	Use this command sequence in the event of a perigee Sun maneuver.	N/A	
Precession	2.	Verify Thruster enable command occurs within STF maneuver window.		
Maneuver:	3.	Verify no eclipse occurs from cat bed heater on through set FC Mode Mission.		
Perigee	4.	, and the state of		
	_	• CATBED_5N_HTR,ON		
	5.	Verify in FC mode Burn and Sun target.		
		SetFcMode burn		
		SetLrTarget ACS_SUN		
	6.	Verify thrust time set to 16 min.		
	_	SetThrustTime 960		
	7.	Verify thruster enable command matches SunMvrBegin time in OEF.		
		SetThrustEnable ENABLE		
		SunMvrBegin		
	8.	Verify 15 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
		to 0, FC mode Mission.		
		SetThrustEnable DISABLE		
		SetHTRCmd CATBED_5N_HTR,OFF		
		SetThrustTime 0		
	l	SetFcMode Mission		1

## IBEX Command Approval Checklist K. Fairchild

Rev 17 Last update 11/05/2022

Battery Cell Balancing	1. 2. 3. 4. 5. 6. 7.	There will be battery cell balancing every 2 out of 3 orbits. Battery cell balancing this orbit? N Verify charging cycle (Long eclipse flag=TRUE) is 90 minutes long.  Verify the first command sets the long eclipse flag to TRUE, the second command sets the flag to FALSE.  Verify P/L is in HVSTANDBY or HVENG. (The DESCENDING_PL1 macro leaves us in HVENG; the DESCENDING_PL2 macro leaves us in HVSTANDBY.)  Verify no charging cycle within 2 hours of ASCENDING or DESCENDING macro execution.  Verify no charging cycle within 15 minutes of maneuver.  Verify no charging cycle during an eclipse.	12/11/22	NGA
Cmd Violation	1.	Review CCVR. If you have any questions Reply All to the ATS Approval email and ask the team.	12/11/22	NGA

Activity	Anomaly Response : Non-nominal burn	Date Completed	Completed By
	If the maneuver has not occurred or the spacecraft pointing as designated by either the star tracker or coarse Sun sensor is off by more than 2 degrees from the expected pointing, an anomaly has occurred.		-
	If the spacecraft is in Contingency state all stored commands are flushed from the command queue. Follow standard anomaly process.		
	If the Sun maneuver did not occur and the spacecraft is in either Science or Housekeeping state follow the steps below.      The Sun maneuver did not occur and the spacecraft is in either Science or Housekeeping state follow the steps below.		
	If a partial Sun maneuver has occurred and the spacecraft is in either Science or Housekeeping state follow the steps below.		
	<ul> <li>Please note that there is no anticipated hardware damage associated with exceeding the 12.5 degree constraint; this constraint is in place because we should not exceed the largest pointing achieved thus far in the mission.</li> </ul>		
	<ol> <li>If the payload is in HVSTANDBY, LVENG or OFF, and there are no commands loaded to bring it to HVSCI voltages, no operational pointing constraints will be violated. Follow standard anomaly process.</li> </ol>		
	2. If the payload is in HVSCI mode or there are uploaded commands to bring the payload to HVSCI mode,	ļ	
	<ul> <li>Determine current off Sun pointing. If the off Sun angle has already exceeded 12.5 degrees, the MOC should notify the MOM and immediately send the DESCENDING command suite in real-time, as described below. If this cannot be</li> </ul>		
	done in the contact where the pointing anomaly was discovered, another contact will be planned as soon as possible to execute these commands.  @CEU_MACRO_EXEC DESCENDING_PL1 (< 1 min)		
	@CEU_MACRO_EXEC DESCENDING_LO (~ 10 min) @CEU_MACRO_EXEC DESCENDING_HI (~ 8 min) @CEU_MACRO_EXEC DESCENDING_PL2 (< 1 min)		
	Please note: The only scenario where hitting 12.5 degrees is expected is when a perigee Sun precession maneuver is completely missed after an inertial apogee maneuver which occurs late in the maneuver window (near apogee + 10 hours).		
	<ul> <li>If current Sun pointing is below 12.5 degrees, the ISOC should input ST Sun- pointing angle into ibex_rotate to determine the Sun angle at the time of</li> </ul>		
	DESCENDING. Sun-Angle at payload DESCENDING  c. If 12.5 degrees is not exceeded by the time of DESCENDING, no payload-specific		
	action is needed for this arc. Follow standard anomaly response process. d. If 12.5 degrees is exceeded by the time of DESCENDING, a new command set	ļ	
	should be sent.  i. The ISOC will generate the new STF which has the DESCENDING commands executing early such that 12.5 degrees is not exceeded while		
	the payload is in HVSCI.  ii. The MOC will create an associated ATS.  iii. Approval is payded by the MOM MOC & ISOC prior to upleed		
	iii.Approval is needed by the MOM, MOC & ISOC prior to upload. iv.The MOC will assess whether an additional pass is needed in order to upload the commands prior to exceeding the 12.5 degree constraint. The		
	onboard DESCENDING commands do not need to be deleted.		