# IBEX Command Approval Checklist <u>Rev 16e removes the Moon</u> in Lo FOV section.

	Rev 16e re	moves the	Moon in Lo	FOV section.
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Orbit	490	Special Ops				
14 R <sub>E</sub> asc. Date/Time	6/6/2020 21:	03:16	15 R <sub>E</sub> asc. Date/Time	6/6/2020 23:11:33		
Apogee	6/11/2020 00:15:13	Maneuver Window Start	6/10/2020 14:21:14	Maneuver Window End	6/11/2020 10:21:16	
Apogee Target		43200000000001 730999999999999	targetY:0.90132	10999999999997		
15 R <sub>E</sub> desc. Date/Time	6/15/2020 02	:22:19	14 R <sub>E</sub> desc. Date/Time	6/15/2020 04:40:25		
Perigee	6/15/2020 12:39:21	Maneuver Window Start	6/15/2020 05:17:47	Maneuver Window Stop	6/15/2020 20:43:19	
Perigee Target	-	90099999999999999999999999999999999999	targetY:0.91290	09999999999996		
Eclipse	No	Eclipse Start		Eclipse End		
Sun Mnvr	No	Apogee/Perigee		Sun Angle at DESCENDING		
Approved Version	IBEX_2020_158_00490a_BUS_v001.scr, IBEX_2020_158_00490a_INST_v001.scr					

Activity		Command Checks	Date Done	Done By
Supporting Materials	1. 2. 3. 4. 5. 6.	IBEX_CrossingTimes_ <date>_v00x.txt on SFTP at /Archive- Incoming/IBEXIncoming/FDG/PredictedEphemeris/Orbit Events/. Orbit Events File on SFTP at /IBEXOutgoing/MOC/Moc-Soc/oef/. Command Constraint Violations Report on SFTP at /IBEXOutgoing/MOC/Moc-Soc/cvr/. Contacts this orbit Orbit_oXXX.txt included in the ATS approval email. Science Tasking File at /Archive-Incoming/IBEXIncoming/MOC/Soc-Moc/stf/. Merged ATS at http://ibex.unh.edu/cgi-bin/ats.cgi.</date>	05/14/20	NGA
Sun Maneuvers	•	Additional contacts <i>should not</i> be planned to support IBEX Sun Precession Maneuvers due to star tracker outages. The standard apogee and perigee contacts should be used to verify that a maneuver has occurred. If it is not possible to plan one of the standard contacts after the star tracker outage is down to 50% and a valid quaternion reading can be made to verify the maneuver, the coarse Sun sensor angle and the thruster pulse count will be used to determine a) whether a maneuver took place, and b) whether the pointing after the maneuver is as expected +/- 2 degrees. The nominal off-Sun pointing constraint is 7.25 degrees. Based on the missed maneuver in orbit	N/A	
	•	114, the payload team has determined that there is no hardware risk associated with off-Sun pointing up to angles of at least 13 degrees. There is a higher background noted in the data starting at around 9.5 degrees off Sun pointing.		
File Input Check	1. 2. 3.	Current OEF inputs are Forecast STF, last orbit's OEF & latest predictive ephemeris. ATS inputs are this orbit's OEF & STF. (And ABS if present.) ATS filename is of the format IBEX_yyyy_doy_o0xxxa_v0zz.scr. where IBEX is capitalized, yyyy is the year, doy is the day of year, xxx is the 3-digit orbit number and zz is the 2-digit version number. Any special operations ATSs will have another designation between the orbit number and version number (i.e. *o0186a_hgc_v001 for the Hi gain curve).	05/14/20 BUS & INST ATS split, so different input files	NGA

E alliana a	1.	Check OEF for eclipses during the orbit.	N1/A	
Eclipses	<ol> <li>2.</li> <li>3.</li> <li>4.</li> </ol>	<ul> <li>Verify long eclipses during the orbit.</li> <li>Verify long eclipse flag start &amp; stop times reflect Ryan Tyler's recommendations based on his eclipse diagnostic tool. Suggestions made by Ryan after the use of this tool trump the general guidelines below. (Please note, specific timing may shift if the recommendations are relative to eclipse timing. For example, it may say set LE flag to false X hours after the end of the eclipse with a given FALSE time suggested. If the eclipse timing changes as the ephemeris becomes more refined, this command time may shift.)</li> <li>Verify no contacts planned during an eclipse. Note: If in conflict, the eclipse diagnostic recommendations will trump the general guidelines below.</li> <li>a. Verify transmitter OFF from 30 minutes before eclipse start through the end of the eclipse.</li> <li>b. For an eclipse where the long eclipse flag is set, schedule a SOH contact directly following the end of the eclipse (or per Ryan's assessment).</li> <li>c. Set the LE flag according to Ryan's assessments.</li> <li>Verify no maneuver planned during an eclipse. Note: If in conflict, the eclipse diagnostic recommendations will trump the general guidelines below.</li> <li>a. Verify no maneuver planned during an eclipse. Note: If in conflict, the eclipse diagnostic recommendations will trump the general guidelines below.</li> </ul>	N/A	NGA
	5.	<ul> <li>b. Verify no maneuver or cat bed heaters are on while the long eclipse flag is set.</li> <li>Verify the following additional constraints (from battery balancing section).</li> <li>a. Verify the first command sets the long eclipse flag to TRUE, the second command sets the flag to FALSE.</li> <li>b. Verify P/L is in HVSTANDBY or HVENG.</li> <li>c. Verify no charging cycle within 2 hours of ASCENDING or DESCENDING macro</li> </ul>		
		execution. This applies to all eclipses, not just moderate or long eclipses.		
_	1.	Verify STX on/off times, downlink rate against <i>Orbit_oXXX.txt</i> file.		
Contact	2.	Verify contacts in the previous ATS have not been duplicated.	05/14/20	NGA
Commands	3.	Verify all currently planned contacts in Orbit_xxx.txt are in the ATS.		
	4.	Verify each contact contains the following 5 commands.		
		SetRelay stx,on		
		<ul> <li>SetDownlink2K (2K, 40K, 64K, 160K, or 320K)</li> </ul>		
		SetBilevelOutputControlReg STXMODE_Strobe,ON		
		SetBilevelOutputControlReg COHERENT,ON		
		SetBleveloupditeointoixeg contraction		
	5.	If contact is near an eclipse		
	5.	<ul> <li>Verify transmitter OFF from 30 minutes before eclipse start through 30 minutes after eclipse end.</li> </ul>		
		b. If additional transmitter constraints exist, they will be captured in Ryan's		
	6.	recommendations. If an APL contact is being used for an SSR Dump, the data rate should be at least 160 ksps & the SSR DUMP_NEW command should be included in the contact commands. Commands in Orange should entry be contributed and the reset this periges. These commands about		
		should only be sent if the SSR pointers need to be reset this perigee. These commands should be separated by 2s and occur 2s after the SSR_DUMP_NEW command.		
		<ul> <li>SetRelay stx,on</li> <li>SetDownlink2K</li> </ul>		
		SetBilevelOutputControlReg STXMODE_Strobe,ON     SetBilevelOutputControlReg CONFERENT ON		
		SetBilevelOutputControlReg COHERENT,ON		
		SetDownlink320K		
		SSR_DUMP_NEW		
	1	SSR_SET_RD_PTR 6500     SSR_SET_WIT_DTR 6500		
		SSR_SET_WRT_PTR 6500		
		SetRelay stx,off		l

SC State Science: arc a	<ol> <li>Transition to Science state will be first command of each ATS (at 14 Re).         <ul> <li>SetScState science \$TIME=2020/06:06:21:04:49</li> </ul> </li> <li>Lo science mode will be the next command (at 14 Re).         <ul> <li>LO_SCIENCE_MODE NORMAL</li> </ul> </li> <li>Verify no transition to Science again at the end of the ATS. The ATS commands go from 14 Re to 14 Re in each orbit.</li> <li>Verify the transition to Science commands for this orbit are not part of the previous ATS using <a href="http://ibex.unh.edu/cgi-bin/ats.cgi">http://ibex.unh.edu/cgi-bin/ats.cgi</a>.</li> <li>Verify that the beginning of this ATS does not overlap with the end of the previous orbit's ATS using <a href="http://ibex.unh.edu/cgi-bin/ats.cgi">http://ibex.unh.edu/cgi-bin/ats.cgi</a>.</li> </ol>	05/14/20	NGA
Payload Mode HVSCI : arc a	<ol> <li>Verify w/ Crossing Times report that it occurs about 15 Re ascending. The arc a ASCENDING commands can start any time at or above 15 Re ascending.</li> <li>Verify commanding takes ~ 24 minutes.         <ul> <li>ASCENDING_PL1 \$TIME=2020/06:06:23:11:42</li> <li>ASCENDING_HI</li> <li>SET_PARAMETER 1, TLM_RATE_SOH</li> <li>SET_PARAMETER 4, HV_STEP_DWELL</li> <li>SET_PARAMETER 3, HV_STEP_FRAC</li> <li>HI_COL_NEG_LVL 1400</li> <li>CEU_HI_CEM_1_LVL 1780</li> <li>CEU_HI_CEM_3_LVL 1780</li> <li>CEU_HI_CEM_3_LVL 1780</li> <li>SET_PARAMETER 0, TLM_RATE_SOH</li> <li>ASCENDING_PL2</li> <li>ASCENDING_PL1</li> <li>ASCENDING_PL1</li> </ul> </li> </ol>	05/14/20	NGA
Payload Mode HVSTANDBY : arc a	<ul> <li>ASCENDING_PL2 \$TIME=2020/06:06:23:35:30</li> <li>Payload DESCENDING commands end 1.5h before thruster enable.</li> <li>Verify commands take ~23 minutes to execute.         <ul> <li>DESCENDING_PL1 \$TIME=2020/06:10:12:25:16</li> <li>DESCENDING_LO</li> <li>ASCENDING_PL2</li> <li>DESCENDING_PL1</li> <li>DESCENDING_PL1</li> <li>DESCENDING_PL1</li> <li>DESCENDING_PL2</li> <li>DESCENDING_PL2</li> <li>DESCENDING_PL2</li> <li>DESCENDING_PL1</li> <li>DESCENDING_PL2</li> <li>STIME=2020/06:10:12:48:06</li> </ul> </li> </ul>	05/14/20	NGA
SC State HK : arc a	1 Spacecraft Housekeeping command occurs 1b before thruster enable	05/14/20	NGA

#### Last update 09/12/2018

Inertial	1.	Use this command sequence if an apogee inertial maneuver is used, otherwise skip to the 'Sun Precession Maneuver : Apogee' sequence below.	05/14/20	NGA
Maneuver : Apogee	2.	Verify Thruster enable command occurs within STF maneuver window.		
1 0	3. ⊿	Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
	4.	<ul> <li>Verify cat bed heaters come on 55 min before burn.</li> <li>CATBED_5N_HTR,ON \$TIME=2020/06:10:13:26:14</li> </ul>		
	5.	Verify Kalman Filter input select is ground command & estimator update is disabled.		
	-	• SetKFInputSelect GND_CMD, 0, 0, 0, 0		
		SetEstUpdateEnables ENABLE, DISABLE		
	6.	Verify in FC mode Burn.		
		SetFcMode burn		
	7.	Compare SetInrDir in ATS with pointing as defined in the Forecast STF & verify the vectors		
		match.		
		<ul> <li>SetInrDir 0.208432,0.901311,0.379731</li> <li>targetX:0.2084320000000001 targetY:0.90131099999999997 </li> </ul>		
		targetZ:0.3797309999999999		
	8.	Verify inertial maneuver chosen.		
	_	SetLrTarget ACS_INERTIAL		
	9.	Verify thrust time set to 11 min.		
		SetThrustTime 660		
	10.	Verify thruster enable command matches RepointingManeuverStart time in OEF.		
		SetThrustEnable ENABLE \$TIME=2020/06:10:14:21:14		
	11	RepointingManeuverStart 2020-06-10T14:21:14  Vorify 10 minutes after thrusters analysis thrusters disabled, act had beaters off thrust time act		
	11.	Verify 10 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set to 0.		
		SetThrustEnable DISABLE \$TIME=2020/06:10:14:31:14		
		SetHTRCmd CATBED_5N_HTR,OFF		
		SetThrustTime 0		
	12.	Verify 25 min after thrusters enabled: Static Z rate set, outage %valid set, FC mode Mission.		
		<ul> <li>SetStaticZrate ESTIMATOR, 0.418 \$TIME=2020/06:10:14:46:24</li> </ul>		
		SetKFInputSelect STA_PCT_VALID, 43,28,33,48		
		SetFcMode Mission		
Sun	1.	Use this command sequence in the event of an apogee Sun maneuver. Verify Thruster enable command occurs within STF maneuver window.	N/A	NGA
Precession	2. 3.	Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
Maneuver : Apogee		Verify cat bed heaters powered on 55 min before thruster enable.		
. Apogee		CATBED_5N_HTR,ON		
	5.	Verify in FC mode Burn and Sun target.		
		SetFcMode burn		
	~	SetLrTarget ACS_SUN		
	6.	Verify thrust time set to 16 min.     SetThrustTime 960		
	7.	Verify thruster enable command matches SunMvrBegin time in OEF.		
	<i>'</i> .	SetThrustEnable ENABLE		
		SunMvrBegin		
	8.	Verify 15 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
		to 0, FC mode Mission.		
		SetThrustEnable DISABLE		
		SetHTRCmd CATBED_5N_HTR,OFF		
		SetThrustTime 0		
	4	SetFcMode Mission		
SC State	1.	<ul> <li>Spacecraft Science commands occur ~1h after thruster Disable.</li> <li>SetScState science \$TIME=2020/06:10:15:36:20</li> </ul>	05/14/20	NGA
Science :		<ul> <li>SetScState science \$110E=2020/06:10:15:36:20</li> <li>LO_SCIENCE_MODE NORMAL</li> </ul>		
arc b	1			

Rev 16e

#### Rev 16e Last update 09/12/2018

Payload Mode HVSCI : arc b	1. 2.	Verify payload ASCENDING commands begin at least 1.5 hours after thruster DISABLE. Verify commands take ~24 minutes to execute. ASCENDING_PL1 \$TIME=2020/06:10:16:06:20 ASCENDING_HI SET_PARAMETER 1, TLM_RATE_SOH SET_PARAMETER 4, HV_STEP_DWELL SET_PARAMETER 3, HV_STEP_FRAC HI_COL_NEG_LVL 1400 CEU_HI_CEM_1_LVL 1780 CEU_HI_CEM_2_LVL 1780 CEU_HI_CEM_3_LVL 1780 CEU_HI_CEM_4_LVL 1900 SET_PARAMETER 0, TLM_RATE_SOH ASCENDING_PL2 ASCENDING_PL2 SET_DING_LO ASCENDING_LO	05/14/20	NGA
Payload Mode HVSTANDBY : arc b	1. 2.	Verify W/ Crossing Times report that it occurs about 15 Re descending. The arc b DESCENDING commands can complete any time at or above 15 Re descending. Verify commands take ~23 minutes to execute. • DESCENDING_PL1 \$TIME=2020/06:15:01:56:18 • DESCENDING_LO • ASCENDING_PL2 • DESCENDING_PL1 • DESCENDING_PL1 • DESCENDING_PL2 • DESCENDING_PL2	05/14/20	NGA
SC State HK : arc b	1.	<ul> <li>Verify with Crossing Times report that Transition to Housekeeping state occurs at 14 Re desc (or an hour before the maneuver if the maneuver occurs less than 1 hour after 14 Re desc).</li> <li>SetScState housekeeping \$TIME=2020/06:15:04:12:01</li> </ul>	05/14/20 10h33m before maneuver due to Star Tracker outage	NGA

Inertial	1.	Use this command sequence if a perigee inertial maneuver is used, otherwise skip to the 'Sun	05/14/20	NGA
Maneuver :		Precession Maneuver : Perigee' sequence below.		
Perigee	2.	Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
	3.	Verify cat bed heaters come on 55 min before burn.		
		• CATBED_5N_HTR,ON \$TIME=2020/06:15:13:49:49		
	4.	Verify in Housekeeping state.		
	5.	Verify Kalman Filter input select is ground command & estimator update is disabled.		
		• SetKFInputSelect GND_CMD, 0, 0, 0, 0		
		SetEstUpdateEnables ENABLE, DISABLE		
	6.	Verify in FC mode Burn.		
		SetFcMode burn		
	7.	Compare SetInrDir in ATS with target vector in the Forecast STF & verify the vectors match.		
	7.	<ul> <li>SetInrDir 0.132901,0.912901,0.385938</li> </ul>		
		targetX:0.132900999999999999999990991targetY:0.91290099999999999999		
		targetZ:0.3859380000000000		
	8.	Verify inertial maneuver chosen.		
	0.	•		
	9.	SetLrTarget ACS_INERTIAL     Verify thrust time set to 11 min.		
	9.	•		
	10	SetThrustTime 660		
	10.	Verify thruster enable command matches RepointingManeuverStart time in OEF.		
		SetThrustEnable ENABLE \$TIME=2020/06:15:14:44:49		
		RepointingManeuverStart 2020-06-15T14:44:49		
	11.	Verify 10 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
		to 0.		
		<ul> <li>SetThrustEnable DISABLE \$TIME=2020/06:15:14:54:49</li> </ul>		
		CATBED_5N_HTR,OFF		
		SetThrustTime 0		
	12.	Verify 25 min after thrusters enabled: Static Z rate set, outage %valid set, FC mode Mission.		
		<ul> <li>SetStaticZrate ESTIMATOR, 0.418 \$TIME=2020/06:15:15:09:59</li> </ul>		
		<ul> <li>SetKFInputSelect STA_PCT_VALID, 43,28,33,48</li> </ul>		
		SetFcMode Mission		
Sun	1.	Use this command sequence in the event of a perigee Sun maneuver.	N/A	NG
Precession	2.	Verify Thruster enable command occurs within STF maneuver window.		
Aneuver :	3.	Verify no eclipse occurs from cat bed heater on through set FC Mode Mission.		
Perigee	4.	Verify cat bed heaters come on 55 min before burn.		
0		• CATBED_5N_HTR,ON		
	5.	Verify in FC mode Burn and Sun target.		
		SetFcMode burn		
		SetLrTarget ACS_SUN		
	6.	Verify thrust time set to 16 min.		
	_	SetThrustTime 960		
	7.	Verify thruster enable command matches SunMvrBegin time in OEF.		
		SetThrustEnable ENABLE		
		• SunMvrBegin		
	8.	Verify 15 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
		to 0, FC mode Mission.		
		SetThrustEnable DISABLE		
		SetHTRCmd CATBED_5N_HTR,OFF		
		SetThrustTime 0		
	1	SetFcMode Mission	1	

Battery Cell Balancing	<ol> <li>There will be battery cell balancing every 2 out of 3 orbits. Battery cell balancing this orbit? N</li> <li>Verify charging cycle (Long eclipse flag=TRUE) is 90 minutes long.</li> <li>Verify the first command sets the long eclipse flag to TRUE, the second command sets the flag to FALSE.</li> <li>Verify P/L is in HVSTANDBY or HVENG.</li> <li>Verify no charging cycle within 2 hours of ASCENDING or DESCENDING macro execution.</li> <li>Verify no charging cycle within 1 hour of maneuver.</li> <li>Verify no charging cycle during an eclipse.</li> </ol>	05/14/20	NGA
Cmd Violation	1. Review CCVR. If you have any questions Reply All to the ATS Approval email and ask the team.	05/14/20	NGA

Activity		Anomaly Response : Non-nominal burn	Date Completed	Completed By
	•	If the maneuver has not occurred or the spacecraft pointing as designated by either the star	_	_
		tracker or coarse Sun sensor is off by more than 2 degrees from the expected pointing, an		
		anomaly has occurred.		
		• If the spacecraft is in Contingency state all stored commands are flushed from the		
		command queue. Follow standard anomaly process.		
		If the Sun maneuver did not occur and the spacecraft is in either Science or		
		Housekeeping state follow the steps below.		
		• If a partial Sun maneuver has occurred and the spacecraft is in either Science or		
		Housekeeping state follow the steps below.		
	•	Please note that there is no anticipated hardware damage associated with exceeding the		
		12.5 degree constraint; this constraint is in place because we should not exceed the largest		
	1	pointing achieved thus far in the mission.		
	1.	If the payload is in HVSTANDBY, LVENG or OFF, and there are no commands loaded to bring it to HVSCI voltages, no experitional pointing constraints will be violated. Follow		
		bring it to HVSCI voltages, no operational pointing constraints will be violated. Follow standard anomaly process.		
	2.	If the payload is in HVSCI mode or there are uploaded commands to bring the payload to		
	2.	HVSCI mode.		
		a. Determine current off Sun pointing. If the off Sun angle has already exceeded 12.5		
		degrees, the MOC should notify the MOM and immediately send the		
		DESCENDING command suite in real-time, as described below. If this cannot be		
		done in the contact where the pointing anomaly was discovered, another contact		
		will be planned as soon as possible to execute these commands.		
		<pre>@CEU_MACRO_EXEC DESCENDING_PL1 (&lt; 1 min)</pre>		
		<pre>@CEU_MACRO_EXEC DESCENDING_LO (~ 10 min)</pre>		
		<pre>@CEU_MACRO_EXEC DESCENDING_HI (~ 8 min)</pre>		
		@CEU_MACRO_EXEC_DESCENDING_PL2 (< 1 min)		
		Please note: The only scenario where hitting 12.5 degrees is expected is when a		
		perigee Sun precession maneuver is completely missed after an inertial apogee		
		maneuver which occurs late in the maneuver window (near apogee + 10 hours).		
		<ul> <li>b. If current Sun pointing is below 12.5 degrees, the ISOC should input ST Sun- pointing angle into ibex rotate to determine the Sun angle at the time of</li> </ul>		
		DESCENDING.		
		Sun-Angle at payload DESCENDING		
		c. If 12.5 degrees is not exceeded by the time of DESCENDING, no payload-specific		
		action is needed for this arc. Follow standard anomaly response process.		
		d. If 12.5 degrees is exceeded by the time of DESCENDING, a new command set		
		should be sent.		
	1	i. The ISOC will generate the new STF which has the DESCENDING		
	1	commands executing early such that 12.5 degrees is not exceeded while		
	1	the payload is in HVSCI.		
	1	ii. The MOC will create an associated ATS.		
	1	iii. Approval is needed by the MOM, MOC & ISOC prior to upload.		
		iv. The MOC will assess whether an additional pass is needed in order to		
		upload the commands prior to exceeding the 12.5 degree constraint. The		
	1	onboard DESCENDING commands do not need to be deleted.		