IBEX Command Approval Checklist <u>Rev 16e removes the Moon</u> in Lo FOV section.

	Rev 16e	removes the	Moon in Lo	FOV section.
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Orbit	458	Special Ops			
14 R _E asc. Date/Time	8/21/2019 14:14:56 15 R _E asc. Date/Time 8/21/2019 16:16:16			16:16	
Apogee	8/25/2019 17:57:41	Maneuver Window Start	8/25/2019 08:02:00	Maneuver Window End	8/26/2019 04:02:02
Apogee Target	-	247099999999999 974000000000001	99 targetY:0.4682	283999999999998	
15 R _E desc. Date/Time	8/29/2019 19:53:05 14 R _E desc. Date/Time 8/29/2019 21:54:13				54:13
Perigee	8/30/2019 06:51:35	Maneuver Window Start	8/29/2019 22:29:04	Maneuver Window Stop	8/30/2019 15:53:38
Perigee Target	-	89030000000000 977000000000000	1 targetY:0.4054	492999999999999999	
Eclipse	No	Eclipse Start		Eclipse End	
Sun Mnvr	No	Apogee/Perigee		Sun Angle at DESCENDING	
Approved Version	IBEX_2019_233	_00458a_v001.s	scr		

Activity		Command Checks	Date Done	Done By
Supporting Materials	1. 2. 3. 4. 5. 6.	IBEX_CrossingTimes_ <date>_v00x.txt on SFTP at /Archive- Incoming/IBEXIncoming/FDG/PredictedEphemeris/Orbit Events/. Orbit Events File on SFTP at /IBEXOutgoing/MOC/Moc-Soc/oef/. Command Constraint Violations Report on SFTP at /IBEXOutgoing/MOC/Moc-Soc/cvr/. Contacts this orbit Orbit_oXXX.txt included in the ATS approval email. Science Tasking File at /Archive-Incoming/IBEXIncoming/MOC/Soc-Moc/stf/. Merged ATS at http://ibex.unh.edu/cgi-bin/ats.cgi.</date>	08/1019	NGA
Sun Maneuvers	•	Additional contacts <i>should not</i> be planned to support IBEX Sun Precession Maneuvers due to star tracker outages. The standard apogee and perigee contacts should be used to verify that a maneuver has occurred. If it is not possible to plan one of the standard contacts after the star tracker outage is down to 50% and a valid quaternion reading can be made to verify the maneuver, the coarse Sun sensor angle and the thruster pulse count will be used to determine a) whether a maneuver took place, and b) whether the pointing after the maneuver is as expected +/- 2 degrees.	N/A	NGA
	•	The nominal off-Sun pointing constraint is 7.25 degrees. Based on the missed maneuver in orbit 114, the payload team has determined that there is no hardware risk associated with off-Sun pointing up to angles of at least 13 degrees. There is a higher background noted in the data starting at around 9.5 degrees off Sun pointing.		
File Input Check	1. 2. 3.	Current OEF inputs are Forecast STF, last orbit's OEF & latest predictive ephemeris. ATS inputs are this orbit's OEF & STF. (And ABS if present.) ATS filename is of the format IBEX_yyyy_doy_o0xxxa_v0zz.scr. where IBEX is capitalized, yyyy is the year, doy is the day of year, xxx is the 3-digit orbit number and zz is the 2-digit version number. Any special operations ATSs will have another designation between the orbit number and version number (i.e. *o0186a_hgc_v001 for the Hi gain curve).	08/1019	NGA

	1.	Check OEF for eclipses during the orbit.		
Eclipses	2. 3. 4. 5.	 Verify long eclipse flag start & stop times reflect Ryan Tyler's recommendations based on his eclipse diagnostic tool. Suggestions made by Ryan after the use of this tool trump the general guidelines below. (Please note, specific timing may shift if the recommendations are relative to eclipse timing. For example, it may say set LE flag to false X hours after the end of the eclipse with a given FALSE time suggested. If the eclipse timing changes as the ephemeris becomes more refined, this command time may shift.) Verify no contacts planned during an eclipse. Note: If in conflict, the eclipse diagnostic recommendations will trump the general guidelines below. a. Verify transmitter OFF from 30 minutes before eclipse start through the end of the eclipse. b. For an eclipse where the long eclipse flag is set, schedule a SOH contact directly following the end of the eclipse (or per Ryan's assessment). c. Set the LE flag according to Ryan's assessments. Verify no maneuver planned during an eclipse. Note: If in conflict, the eclipse diagnostic recommendations will trump the general guidelines below. a. Verify no maneuver or cat bed heaters on from 3 hours before eclipse start through 3 hours after eclipse end. b. Verify no maneuver or cat bed heaters are on while the long eclipse flag is set. Verify the following additional constraints (from battery balancing section). a. Verify the first command sets the long eclipse flag to TRUE, the second command sets the flag to FALSE. b. Verify P/L is in HVSTANDBY or HVENG. 	N/A	NGA
		c. Verify no charging cycle within 2 hours of ASCENDING or DESCENDING macro		
		execution.		
	1.	This applies to all eclipses, not just moderate or long eclipses. Verify STX on/off times, downlink rate against Orbit_oXXX.txt file.		
Contact	1. 2.	Verify contacts in the previous ATS have not been duplicated.	08/1019	NGA
Commands	3.	Verify all currently planned contacts in Orbit_xxx.txt are in the ATS.		
	4.	Verify each contact contains the following 5 commands.		
		SetRelay stx,on		
		 SetDownlink2K (2K, 40K, 64K, 160K, or 320K) 		
		SetBilevelOutputControlReg STXMODE_Strobe,ON		
		SetBilevelOutputControlReg COHERENT,ON		
		SetRelay stx,off		
	5.	If contact is near an eclipse		
	_	a. Verify transmitter OFF from 30 minutes before eclipse start through 30 minutes after		
		eclipse end.		
		b. If additional transmitter constraints exist, they will be captured in Ryan's		
	c	recommendations. If an APL contact is being used for an SSR Dump, the data rate should be at least 160 ksps & the		
	6.	SSR DUMP NEW command should be included in the contact commands. Commands in Orange		
		should only be sent if the SSR pointers need to be reset this perigee. These commands should		
		be separated by 2s and occur 2s after the SSR_DUMP_NEW command.		
		SetRelay stx,on		
		SetDownlink2K		
		 SetBilevelOutputControlReg STXMODE_Strobe,ON 		
		 SetBilevelOutputControlReg COHERENT,ON 		
		SetDownlink320K		
		• SSR_DUMP_NEW		
		• SSR_SET_RD_PTR 6500		
		• SSR_SET_WRT_PTR 6500		
		SetRelay stx,off		

SC State Science: arc a	 Transition to Science state will be first command of each ATS (at 14 Re). SetScState science \$TIME=2019/08:21:14:15:09 Lo science mode will be the next command (at 14 Re). LO_SCIENCE_MODE NORMAL Verify no transition to Science again at the end of the ATS. The ATS commands go from 14 Re to 14 Re in each orbit. Verify the transition to Science commands for this orbit are not part of the previous ATS using http://ibex.unh.edu/cgi-bin/ats.cgi. 	08/1019	NGA
	 Verify that the beginning of this ATS does not overlap with the end of the previous orbit's ATS 		
	using http://ibex.unh.edu/cgi-bin/ats.cgi.		
Payload Mode HVSCI : arc a	 Verify w/ Crossing Times report that it occurs about 15 Re ascending. The arc a ASCENDING commands can start any time at or above 15 Re ascending. Verify commanding takes ~ 24 minutes. ASCENDING_PL1 \$TIME=2019/08:21:16:16:23 ASCENDING_HI SET_PARAMETER 1, TLM_RATE_SOH SET_PARAMETER 4, HV_STEP_DWELL SET_PARAMETER 3, HV_STEP_FRAC HI_COL_NEG_LVL 1400 CEU_HI_CEM_1_LVL 1780 CEU_HI_CEM_3_LVL 1780 CEU_HI_CEM_4_LVL 1900 SET_PARAMETER 0, TLM_RATE_SOH ASCENDING_PL2 ASCENDING_PL1 ASCENDING_LO 	08/1019	NGA
	• ASCENDING_PL2 \$TIME=2019/08:21:16:40:11		
Payload Mode HVSTANDBY : arc a	 Payload DESCENDING commands end 1.5h before thruster enable. Verify commands take ~23 minutes to execute. DESCENDING_PL1 \$TIME=2019/08:25:06:06:02 DESCENDING_LO ASCENDING_PL2 	08/1019	NGA
	DESCENDING_PL1 DESCENDING_HI DESCENDING_PL2 \$TIME=2019/08:25:06:28:52 1. Spacecraft Housekeeping command occurs 1h before thruster enable.		
SC State HK : arc a	• SetScState housekeeping \$TIME=2019/08:25:07:02:02	08/1019	NGA

Last update 09/12/2018

Inertial	1.	Use this command sequence if an apogee inertial maneuver is used, otherwise skip to the 'Sun Precession Maneuver : Apogee' sequence below.	08/1019	NGA
Maneuver :	2.	Verify Thruster enable command occurs within STF maneuver window.		
Apogee	3.	Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
	4.	Verify cat bed heaters come on 55 min before burn.		
		 CATBED_5N_HTR,ON \$TIME=2019/08:25:07:07:00 		
	5.	Verify Kalman Filter input select is ground command & estimator update is disabled.		
		 SetKFInputSelect GND_CMD, 0, 0, 0, 0 		
		SetEstUpdateEnables ENABLE, DISABLE		
	6.	Verify in FC mode Burn.		
		SetFcMode burn		
	7.	Compare SetInrDir in ATS with pointing as defined in the Forecast STF & verify the vectors		
		match.		
		• SetInrDir -0.862471,0.468284,0.191974		
		targetX:-0.86247099999999999 targetY:0.4682839999999998		
	~	targetZ:0.1919740000000001		
	8.	Verify inertial maneuver chosen.		
	0	SetLrTarget ACS_INERTIAL Verify thrust time set to 11 min.		
	9.	SetThrustTime 660		
	10	Verify thruster enable command matches RepointingManeuverStart time in OEF.		
	10.	SetThrustEnable ENABLE \$TIME=2019/08:25:08:02:00		
		 RepointingManeuverStart 2019-08-25T08:02:00 		
	11	Verify 10 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
	11.	to 0.		
		SetThrustEnable DISABLE \$TIME=2019/08:25:08:12:00		
		SetHTRCmd CATBED_5N_HTR,OFF		
		SetThrustTime 0		
	12.	Verify 25 min after thrusters enabled: Static Z rate set, outage %valid set, FC mode Mission.		
		• SetStaticZrate ESTIMATOR, 0.418 \$TIME=2019/08:25:08:27:10		
		SetKFInputSelect STA_PCT_VALID, 43,28,33,48		
		SetFcMode Mission		
Sun	1.	Use this command sequence in the event of an apogee Sun maneuver.	N/A	NGA
Precession	2.	Verify Thruster enable command occurs within STF maneuver window.	11/7	NOA
Maneuver	3.	Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
: Apogee	4.	Verify cat bed heaters powered on 55 min before thruster enable.		
1 0		CATBED_5N_HTR,ON		
	5.	Verify in FC mode Burn and Sun target.		
		SetFcMode burn		
		SetLrTarget ACS_SUN		
	6.	Verify thrust time set to 16 min.		
	-	SetThrustTime 960		
	7.	Verify thruster enable command matches SunMvrBegin time in OEF.		
		SetThrustEnable ENABLE		
	~	SunMvrBegin		
	8.	Verify 15 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set to 0, FC mode Mission.		
		SetThrustEnable DISABLE		
		SetHTRCmd CATBED 5N HTR,OFF		
		SetThrustTime 0		
		SetTrustTime 0 SetFcMode Mission		
	1.	Set CMode Mission Spacecraft Science commands occur ~1h after thruster Disable.		
SC State	1.	SetScState science \$TIME=2019/08:25:09:17:06	08/1019	NGA
Science :		 SetScState Science \$11ME=2019/08:25:09:17:06 LO_SCIENCE_MODE NORMAL 		
arc b	1	LO_SCIENCE_IVIDE NORIVIAL		

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Payload Mode HVSCI : arc b	1. 2.	Verify payload ASCENDING commands begin at least 1.5 hours after thruster DISABLE. Verify commands take ~24 minutes to execute. ASCENDING_PL1 \$TIME=2019/08:25:09:47:06 ASCENDING_HI SET_PARAMETER 1, TLM_RATE_SOH SET_PARAMETER 4, HV_STEP_DWELL SET_PARAMETER 3, HV_STEP_FRAC HI_COL_NEG_LVL 1400 CEU_HI_CEM_1_LVL 1780 CEU_HI_CEM_2_LVL 1780 CEU_HI_CEM_3_LVL 1780 CEU_HI_CEM_4_LVL 1900 SET_PARAMETER 0, TLM_RATE_SOH ASCENDING_PL2 ASCENDING_PL2 SET_PARAMETER 0, TLM_RATE_S0H ASCENDING_PL2 \$TIME=2019/08:25:10:10:54	08/1019	NGA
Payload Mode HVSTANDBY : arc b	1. 2.	 Verify w/ Crossing Times report that it occurs about 15 Re descending. The arc b DESCENDING commands can complete any time at or above 15 Re descending. Verify commands take ~23 minutes to execute. DESCENDING_PL1 \$TIME=2019/08:29:19:27:02 DESCENDING_LO ASCENDING_PL2 DESCENDING_PL1 DESCENDING_HI DESCENDING_PL2 \$TIME=2019/08:29:19:49:52 	08/1019	NGA
SC State HK : arc b	1.	Verify with Crossing Times report that Transition to Housekeeping state occurs at 14 Re desc (or an hour before the maneuver if the maneuver occurs less than 1 hour after 14 Re desc). • SetScState housekeeping \$TIME=2019/08:29:21:24:15	08/1019	NGA

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Inertial Maneuver :	1.	Use this command sequence if a perigee inertial maneuver is used, otherwise skip to the 'Sun Precession Maneuver : Perigee' sequence below.	08/1019	NGA
Perigee	2.	Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
gee		Verify cat bed heaters come on 55 min before burn.		
	0.	• CATBED_5N_HTR,ON \$TIME=2019/08:29:21:34:04		
	4.	Verify in Housekeeping state.		
	 5.	Verify Kalman Filter input select is ground command & estimator update is disabled.		
	5.			
		SetKFInputSelect GND_CMD, 0, 0, 0, 0		
	c	SetEstUpdateEnables ENABLE, DISABLE		
	6.	Verify in FC mode Burn.		
	_	SetFcMode burn		
	7.	Compare SetInrDir in ATS with target vector in the Forecast STF & verify the vectors match.		
		• SetInrDir -0.898903,0.405493,0.165977		
		targetX:-0.8989030000000001 targetY:0.405492999999999999 targetZ:0.16597700000000001		
	8.	Verify inertial maneuver chosen.		
		SetLrTarget ACS_INERTIAL		
	9.	Verify thrust time set to 11 min.		
		SetThrustTime 660		
	10	Verify thruster enable command matches RepointingManeuverStart time in OEF.		
	10.	SetThrustEnable ENABLE \$TIME=2019/08:29:22:29:04		
		 RepointingManeuverStart 2019-08-29T22:29:04 		
	11	Verify 10 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
	11.	to 0.		
		SetThrustEnable DISABLE \$TIME=2019/08:29:22:29:04		
		CATBED_5N_HTR,OFF		
	10	SetThrustTime 0 Varify 25 min ofter thrustore enabled: Static 7 rate act, outpage 9/valid act, EC mode Mission		
	12.	Verify 25 min after thrusters enabled: Static Z rate set, outage %valid set, FC mode Mission.		
		• SetStaticZrate ESTIMATOR, 0.418 \$TIME=2019/08:29:22:54:14		
		SetKFInputSelect STA_PCT_VALID, 43,28,33,48		
		SetFcMode Mission		
Sun	1.	Use this command sequence in the event of a perigee Sun maneuver.	N/A	NGA
Precession	2.	Verify Thruster enable command occurs within STF maneuver window.		
Maneuver :	3.	Verify no eclipse occurs from cat bed heater on through set FC Mode Mission.		
Perigee	4.	Verify cat bed heaters come on 55 min before burn.		
	F	CATBED_5N_HTR,ON Verify in EC mode Burn and Sun target		
	5.	Verify in FC mode Burn and Sun target. SetFcMode burn		
	c	SetLrTarget ACS_SUN Verify thrust time set to 16 min.		
	6.	SetThrustTime 960		
	7.	Verify thruster enable command matches SunMvrBegin time in OEF.		
	/ ·	SetThrustEnable ENABLE		
		SunMvrBegin		
	8.	Verify 15 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
	0.	to 0, FC mode Mission.		
		SetThrustEnable DISABLE		
		SetHTRCmd CATBED_5N_HTR,OFF		
		SetThrustTime 0		
		SetFrMdstrine 0 SetFcMode Mission		
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Battery Cell Balancing	 There will be battery cell balancing every 2 out of 3 orbits. Battery cell balancing this orbit? N Verify charging cycle (Long eclipse flag=TRUE) is 90 minutes long. Verify the first command sets the long eclipse flag to TRUE, the second command sets the flag to FALSE. Verify P/L is in HVSTANDBY or HVENG. Verify no charging cycle within 2 hours of ASCENDING or DESCENDING macro execution. Verify no charging cycle within 1 hour of maneuver. Verify no charging cycle during an eclipse. 	08/1019	NGA
Cmd Violation	1. Review CCVR. If you have any questions Reply All to the ATS Approval email and ask the team.	08/1019	NGA

Activity		Anomaly Response : Non-nominal burn	Date Completed	Completed By
	•	If the maneuver has not occurred or the spacecraft pointing as designated by either the star	_	_
		tracker or coarse Sun sensor is off by more than 2 degrees from the expected pointing, an		
		anomaly has occurred.		
		• If the spacecraft is in Contingency state all stored commands are flushed from the		
		command queue. Follow standard anomaly process.		
		If the Sun maneuver did not occur and the spacecraft is in either Science or		
		Housekeeping state follow the steps below.		
		• If a partial Sun maneuver has occurred and the spacecraft is in either Science or		
		Housekeeping state follow the steps below.		
	•	Please note that there is no anticipated hardware damage associated with exceeding the		
		12.5 degree constraint; this constraint is in place because we should not exceed the largest		
	1	pointing achieved thus far in the mission.		
	1.	If the payload is in HVSTANDBY, LVENG or OFF, and there are no commands loaded to bring it to HVSCI voltages, no experitional pointing constraints will be violated. Follow		
		bring it to HVSCI voltages, no operational pointing constraints will be violated. Follow standard anomaly process.		
	2.	If the payload is in HVSCI mode or there are uploaded commands to bring the payload to		
	2.	HVSCI mode.		
		a. Determine current off Sun pointing. If the off Sun angle has already exceeded 12.5		
		degrees, the MOC should notify the MOM and immediately send the		
		DESCENDING command suite in real-time, as described below. If this cannot be		
		done in the contact where the pointing anomaly was discovered, another contact		
		will be planned as soon as possible to execute these commands.		
		<pre>@CEU_MACRO_EXEC DESCENDING_PL1 (< 1 min)</pre>		
		<pre>@CEU_MACRO_EXEC DESCENDING_LO (~ 10 min)</pre>		
		<pre>@CEU_MACRO_EXEC DESCENDING_HI (~ 8 min)</pre>		
		@CEU_MACRO_EXEC_DESCENDING_PL2 (< 1 min)		
		Please note: The only scenario where hitting 12.5 degrees is expected is when a		
		perigee Sun precession maneuver is completely missed after an inertial apogee		
		maneuver which occurs late in the maneuver window (near apogee + 10 hours).		
		 b. If current Sun pointing is below 12.5 degrees, the ISOC should input ST Sun- pointing angle into ibex rotate to determine the Sun angle at the time of 		
		DESCENDING.		
		Sun-Angle at payload DESCENDING		
		c. If 12.5 degrees is not exceeded by the time of DESCENDING, no payload-specific		
		action is needed for this arc. Follow standard anomaly response process.		
		d. If 12.5 degrees is exceeded by the time of DESCENDING, a new command set		
		should be sent.		
	1	i. The ISOC will generate the new STF which has the DESCENDING		
	1	commands executing early such that 12.5 degrees is not exceeded while		
	1	the payload is in HVSCI.		
	1	ii. The MOC will create an associated ATS.		
	1	iii. Approval is needed by the MOM, MOC & ISOC prior to upload.		
		iv. The MOC will assess whether an additional pass is needed in order to		
		upload the commands prior to exceeding the 12.5 degree constraint. The		
	1	onboard DESCENDING commands do not need to be deleted.		