## IBEX Command Approval Checklist Rev 16b incorporates post-ST anomaly changes and resetting the SSR pointers in an APL contact.

Orbit	343	Special Ops	None			
14 R <sub>E</sub> asc. Date/Time	10/09 18:28		15 R <sub>E</sub> asc. Date/Time	10/09 20:00		
Apogee	10/13 18:15	Maneuver Window Start	10-13T08:15	Maneuver Window End	10-14T04:15	
Apogee Target	_	008899999999999 33467000000000	06 targetY:-0.2819880000000002			
15 R <sub>E</sub> desc. Date/Time	10/17 17:31		14 R <sub>E</sub> desc. Date/Time	10/17 19:09		
Perigee	10/18 05:10	Maneuver Window Start	10-17T19:38	Maneuver Window Stop		
Perigee Target		234149999999999 51449000000000	99 targetY:-0.34 001	481929999999999	7	
Eclipse	None	Eclipse Start	N/A	Eclipse End N/A		
Sun Mnvr	None	Apogee/Perigee	N/A	Sun Angle at DESCENDING N/A		
Approved Version	IBEX_2016_283_		scr			

Activity	Command Checks	Date Done	Done By
Supporting Materials	<ol> <li>IBEX_CrossingTimes_<date>_v00x.txt on SFTP at /IBEX/fdg/PredictedEphemeris/Orbit Events/.</date></li> <li>Orbit Events File on SFTP at /IBEX/moc/Moc-Soc/oef/.</li> <li>Command Constraint Violations Report on SFTP at /IBEX/moc/Moc-Soc/cvr/.</li> <li>Contacts this orbit Orbit_oXXX.txt included in the ATS approval email.</li> <li>Science Tasking File at /IBEX/moc/Soc-Moc/stf/.</li> <li>Merged ATS at <a href="http://ibex.unh.edu/cgi-bin/ats.cgi">http://ibex.unh.edu/cgi-bin/ats.cgi</a>.</li> </ol>		
Sun Maneuvers	<ul> <li>Additional contacts should not be planned to support IBEX Sun Precession Maneuvers due to star tracker outages. The standard apogee and perigee contacts should be used to verify that a maneuver has occurred. If it is not possible to plan one of the standard contacts after the star tracker outage is down to 50% and a valid quaternion reading can be made to verify the maneuver, the coarse Sun sensor angle and the thruster pulse count will be used to determine a) whether a maneuver took place, and b) whether the pointing after the maneuver is as expected +/- 2 degrees.</li> </ul>		
	The nominal off-Sun pointing constraint is 7.25 degrees. Based on the missed maneuver in orbit 114, the payload team has determined that there is no hardware risk associated with off-Sun pointing up to angles of at least 13 degrees. There is a higher background noted in the data starting at around 9.5 degrees off Sun pointing.		
File Input Check	<ol> <li>Current OEF inputs are Forecast STF, last orbit's OEF &amp; latest ephemeris.</li> <li>ATS inputs are this orbit's OEF &amp; STF. (And ABS if present.)</li> <li>ATS filename is of the format IBEX_yyyy_doy_o0xxxa_v0zz.scr.         where IBEX is capitalized, yyyy is the year, doy is the day of year, xxx is the 3-digit orbit number and zz is the 2-digit version number.         Any special operations ATSs will have another designation between the orbit number and version number (i.e. *o0186a_hgc_v001 for the Hi gain curve).</li> </ol>	09/28/2016	CLR

	1.	Check OEF for eclipses during the orbit.	,-	
Eclipses	3.	Verify long eclipse flag start & stop times reflect Ryan Tyler's recommendations based on his eclipse diagnostic tool. Suggestions made by Ryan after the use of this tool trump the general guidelines below. (Please note, specific timing may shift if the recommendations are relative to eclipse timing. For example, it may say set LE flag to false X hours after the end of the eclipse with a given FALSE time suggested. If the eclipse timing changes as the ephemeris becomes more refined, this command time may shift.)  Verify no contacts planned during an eclipse. Note: If in conflict, the eclipse diagnostic recommendations will trump the general guidelines below.  a Verify transmitter OFF from 30 minutes before eclipse start through the end of the eclipse.  b For an eclipse where the long eclipse flag is set, schedule a SOH contact directly	N/A	N/A
		following the end of the eclipse (or per Ryan's assessment).		
	4.	c Set the LE flag according to Ryan's assessments.  Verify no maneuver planned during an eclipse. Note: If in conflict, the eclipse diagnostic recommendations will trump the general guidelines below.  a Verify no maneuver or cat bed heaters on from 3 hours before eclipse start through 3 hours after eclipse end.		
	5.	<ul> <li>b Verify no maneuver or cat bed heaters are on while the long eclipse flag is set.</li> <li>Verify the following additional constraints (from battery balancing section).</li> <li>a Verify the first command sets the long eclipse flag to TRUE, the second command sets the flag to FALSE.</li> <li>b Verify P/L is in HVSTANDBY or HVENG.</li> <li>c Verify no charging cycle within 2 hours of ASCENDING or DESCENDING macro execution.</li> </ul>		
		This applies to all eclipses, not just moderate or long eclipses.		
Moon In Lo FOV	1.	Check OEF for Moon in Lo FOV events.  • MoonInLoFovStart  • MoonInLoFovStop	N/A	N/A
	2.	Check for corresponding Moon in Lo FOV start commands in ATS (timing will not be exact).  • PMT_LVL 300  • IF_STAR_ADJ 0  • Note: if the Moon is closer than 30Re, the PMT will be set to 250. The distance to the moon can be found in the STF.		
	3.	Check for corresponding Moon in Lo FOV stop commands.  IF_STAR_ADJ 250  PMT_LVL 800  Note: if the Moon is still in the FOV at the time of DESCENDING, no Moon in Lo FOV stop commands will be present in ATS. The values are reset to the default at next set of ASCENDING macros.		
	4.	If Moon in Lo FOV starts in arc a & ends in arc b, check Moon in Lo FOV Start commands resent		
	5.	after apogee ASCENDING commands.  If Moon in Lo FOV starts within apogee HVSTANDBY period, check Moon in Lo FOV Start commands sent after apogee ASCENDING commands.		

Contact	4	Fach contact has 5 commands	00/20/2016	CL D
Contact Commands	1.	Each contact has 5 commands.	09/28/2016	CLR
Commands	1.	Verify STX on/off times, downlink rate against <i>Orbit_oXXX.txt</i> file.		
	2.	Verify contacts in the previous ATS have not been duplicated.		
	3.	Verify all currently planned contacts in Orbit_xxx.txt are in the ATS.		
	4.	Verify each contact contains the following 5 commands.		
		SetRelay stx,on		
		• SetDownlink2K (2K, 40K, 64K, 160K, or 320K)		
ļ		SetBilevelOutputControlReg STXMODE_Strobe,ON     SetBilevelOutputControlReg STXMODE_Strobe,ON		
		SetBilevelOutputControlReg COHERENT,ON     SetBilevelOutputControlReg COHERENT,ON		
	_	SetRelay stx,off  If contact is near an eclipse		
	5.			
		<ul> <li>a. Verify transmitter OFF from 30 minutes before eclipse start through 30 minutes after eclipse end.</li> </ul>		
		b. If additional transmitter constraints exist, they will be captured in Ryan's recommendations.		
	6.	If an APL contact is being used for an SSR Dump, the data rate should be at least 160 ksps & the SSR		
	0.	DUMP_NEW command should be included in the contact commands. Commands in Orange should only		
		be sent if the SSR pointers need to be reset this perigee. These commands should be separated by 2s and		
		occur 2s after the SSR_DUMP_NEW command.		
		SetRelay stx,on		
		SetDownlink2K		
		SetBilevelOutputControlReg STXMODE_Strobe,ON		
		SetBilevelOutputControlReg COHERENT,ON		
		SetDownlink320K		
		SSR_DUMP_NEW		
		SSR_SET_RD_PTR 6500		
		SSR_SET_WRT_PTR 6500		
		SetRelay stx,off		
SC State	1.	Transition to Science state will be first command of each ATS (at 14 Re).	09/28/2016	CLR
Science:		SetScState science 10:09:18:29		
arc a	2.	Lo science mode will be the next command (at 14 Re).		
		• LO_SCIENCE_MODE NORMAL		
	3.	Verify no transition to Science again at the end of the ATS. The ATS commands go from 14 Re		
	,	to 14 Re in each orbit.		
	4.	Verify the transition to Science commands for this orbit are not part of the previous ATS using		
	5.	http://ibex.unh.edu/cgi-bin/ats.cgi.  Verify that the beginning of this ATS does not overlap with the end of the previous orbit's ATS		
	٥.	using http://ibex.unh.edu/cgi-bin/ats.cgi.		
	1.	Verify w/ Crossing Times report that it occurs about 15Re ascending. The arc a ASCENDING		
Payload		commands can start any time at or above 15Re ascending.	09/28/2016	CLR
Mode	2.	Verify commanding takes ~ 24 minutes.		
HVSCI: arc a		• ASCENDING_PL1 10:09:20:00		
aica		ASCENDING_HI		
		SET_PARAMETER 1, TLM_RATE_SOH		
		SET_PARAMETER 4, HV_STEP_DWELL		
		SET_PARAMETER 3, HV_STEP_FRAC		
		HI_COL_NEG_LVL 1400		
		CEU_HI_CEM_1_LVL 1780		
		CEU_HI_CEM_2_LVL 1780		
		CEU_HI_CEM_3_LVL 1780		
		CEU_HI_CEM_4_LVL 1900		
		SET_PARAMETER 0, TLM_RATE_SOH		
		ASCENDING_PL2		
		ASCENDING_PL1		
		ASCENDING_LO		
		• ASCENDING_PL2 10:09:20:24		

Payload Mode		Payload DESCENDING commands end 1.5h before thruster enable.     DESCENDING_PL1 10:13:06:52	09/28/2016	CLR
HVSTANDBY : arc a		DESCENDING_LO		
. aic a		ASCENDING_PL2		
		DESCENDING_PL1		
		DESCENDING_HI		
	_	DESCENDING_PL2 10:13:07:15  On a serial black discrete and a serial black discrete and black discrete a		
SC State HK	1.	Spacecraft Housekeeping command occurs 1h before thruster enable.	09/28/2016	CLR
: arc a		SetScState housekeeping 10:13:07:48		
Inertial	1.	Use this command sequence if an apogee inertial maneuver is used, otherwise skip to the 'Sun	09/28/2016	CLR
Maneuver :		Precession Maneuver : Apogee' sequence below.	09/20/2010	OLIX
Apogee		Verify Thruster enable command occurs within STF maneuver window.		
		Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
	4.	Verify cat bed heaters come on 55 min before burn.  • CATBED 5N HTR,ON 10:13:07:53		
	5.	Verify Kalman Filter input select is ground command & estimator update is disabled.		
	Э.	SetKFInputSelect GND CMD, 0, 0, 0, 0		
		SetEstUpdateEnables ENABLE, DISABLE		
	6.	Verify in FC mode Burn.		
	0.	SetFcMode burn		
	7.	Compare SetInrDir in ATS with pointing as defined in the Forecast STF & verify the vectors match.		
		• SetInrDir -0.950089,-0.281988,-0.133467		
		•  targetX:-0.9500889999999996 targetY:-0.2819880000000002		
		targetZ:-0.1334670000000000		
	8.	Verify inertial maneuver chosen.		
		SetLrTarget ACS_INERTIAL		
	9.	Verify thrust time set to 11 min.		
		SetThrustTime 660		
	10.	Verify thruster enable command matches RepointingManeuverStart time in OEF.		
		SetThrustEnable ENABLE 10:13:08:48		
		RepointingManeuverStart 10-13T08:48		
	11.	Verify 10 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
		to 0.		
		SetThrustEnable DISABLE 10:13:08:58		
		SetHTRCmd CATBED_5N_HTR,OFF		
	40	SetThrustTime 0  Verify OF this of the thrust are people of Ottobard Total and To		
	12.	Verify 25 min after thrusters enabled: Static Z rate set, outage %valid set, FC mode Mission.		
		SetStaticZrate ESTIMATOR, 0.418 10:13:09:14		
		SetKFInputSelect STA_PCT_VALID, 43,28,33,48		
		SetFcMode Mission		

	Ι.		<del>                                     </del>	
Sun	1.	Use this command sequence in the event of an apogee Sun maneuver.	N/A	N/A
Precession	2.	Verify Thruster enable command occurs within STF maneuver window.	'	
Maneuver		Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
: Apogee	4.	Verify cat bed heaters powered on 55 min before thruster enable.		
. Apogoo		CATBED_5N_HTR,ON		
	5.	Verify in FC mode Burn and Sun target.		
		SetFcMode burn		
		SetLrTarget ACS_SUN		
	6.	Verify thrust time set to 16 min.		
	0.	SetThrustTime 960		
	7			
	7.	Verify thruster enable command matches SunMvrBegin time in OEF.		
		SetThrustEnable ENABLE		
		SunMvrBegin		
	8.	Verify 15 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
		to 0, FC mode Mission.		
		SetThrustEnable DISABLE		
		SetHTRCmd CATBED 5N HTR,OFF		
		SetThrustTime 0		
		SetFcMode Mission		
	1.	Spacecraft Science commands occur ~1h after thruster Disable.		
SC State	١.	•	09/28/2016	CLR
Science:		• SetScState science 10:13:10:03		
arc b		SCIENCE_MODE NORMAL		
	1.	Verify payload ASCENDING commands begin at least 1.5 hours after thruster DISABLE.		2: 5
Payload	2.	Verify commands take ~24 minutes to execute.	09/28/2016	CLR
Mode		• ASCENDING_PL1 10:13:10:33		
HVSCI:		ASCENDING_HI		
arc b				
		SET_PARAMETER 1, TLM_RATE_SOH		
		SET_PARAMETER 4, HV_STEP_DWELL		
		SET_PARAMETER 3, HV_STEP_FRAC		
		HI_COL_NEG_LVL 1400		
		CEU_HI_CEM_1_LVL 1780		
		• CEU_HI_CEM_2_LVL 1780		
		• CEU_HI_CEM_3_LVL 1780		
		CEU_HI_CEM_4_LVL 1900     SET_PARAMETER 0. TIMA PATE COU		
		SET_PARAMETER 0, TLM_RATE_SOH		
		ASCENDING_PL2		
		ASCENDING_PL1		
		ASCENDING_LO		
		• ASCENDING_PL2 10:13:10:57		
Payload	1.	Verify w/ Crossing Times report that it occurs about 15 Re descending. The arc b DESCENDING	00/00/0045	01.5
Mode	'	commands can complete any time at or above 15Re descending.	09/28/2016	CLR
HVSTANDBY	2	Verify commands take ~23 minutes to execute.		
: arc b	١	DESCENDING_PL1 10:17:17:05		
. 4.0 0		DESCENDING_FET 10.17.17.05      DESCENDING_LO		
	1	_		
	1	ASCENDING PL2	1	
		• DESCENDING_PL1		
		DESCENDING_PL1		
		<ul> <li>DESCENDING_PL1</li> <li>DESCENDING_HI</li> <li>DESCENDING_PL2 10:17:17:27</li> </ul>	2015-1-	
SC State HK	1.	DESCENDING_PL1     DESCENDING_HI     DESCENDING_PL2 10:17:17:27  Verify with Crossing Times report that Transition to Housekeeping state occurs at 14 Re desc (or	09/28/2016	CLR
SC State HK : arc b	1.	<ul> <li>DESCENDING_PL1</li> <li>DESCENDING_HI</li> <li>DESCENDING_PL2 10:17:17:27</li> </ul>	09/28/2016	CLR

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Inertial	1.	Use this command sequence if a perigee inertial maneuver is used, otherwise skip to the 'Sun	09/28/2016	CLR
Maneuver :		Precession Maneuver : Perigee' sequence below.		
Perigee		Verify no eclipse occurs from cat bed heater on through set FC mode Mission.		
	3.	Verify cat bed heaters come on 55 min before burn.		
		• CATBED_5N_HTR,ON 10:17:18:43		
	4.	Verify in Housekeeping state.		
	5. 6.	Verify Kalman Filter input select is ground command & estimator update is disabled.		
	О.	SetKFInputSelect GND_CMD, 0, 0, 0, 0		
	7	SetEstUpdateEnables ENABLE, DISABLE  Verify in EC mode Burn		
	7.	Verify in FC mode Burn.  • SetFcMode burn		
	0	Compare SetInrDir in ATS with target vector in the Forecast STF & verify the vectors match.		
	8.	SetInrDir -0.923415,-0.348193,-0.161449		
		, , , , , , , , , , , , , , , , , , ,		
	9.	•  targetX:-0.923414999999999 targetY:-0.348192999999997 targetZ:-0.16144900000000001 Verify inertial maneuver chosen.		
	9.			
	10	SetLrTarget ACS_INERTIAL  Verify thrust time set to 11 min.		
	10.	SetThrustTime 660		
	11	Verify thruster enable command matches RepointingManeuverStart time in OEF.		
	11.	SetThrustEnable ENABLE 10:17:19:38		
		RepointingManeuverStart 10-17T19:38		
	12	Verify 10 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
	12.	to 0.		
	•	SetThrustEnable DISABLE 10:17:19:48		
		<ul> <li>CATBED_5N_HTR,OFF</li> <li>SetThrustTime 0</li> </ul>		
	12	Verify 25 min after thrusters enabled: Static Z rate set, outage %valid set, FC mode Mission.		
	13.	SetStaticZrate ESTIMATOR, 0.418 10:17:20:03		
		SetStaticZfate ESTIMATOR, 0.418 10.17.20.03     SetKFInputSelect STA PCT VALID, 43,28,33,48		
		SetRemputSelect STA_FCI_VALID, 43,26,33,46      SetFcMode Mission		
	_			
Sun	1. 2.	Use this command sequence in the event of a perigee Sun maneuver.  Verify Thruster enable command occurs within STF maneuver window.	N/A	N/A
Precession		Verify no eclipse occurs from cat bed heater on through set FC Mode Mission.		
Maneuver:	3. 4.	Verify had eclipse occurs from cat bed fleater on through set if a wode Mission.  Verify cat bed heaters come on 55 min before burn.		
Perigee	٠.	CATBED_5N_HTR,ON		
	5.	Verify in FC mode Burn and Sun target.		
	٥.	SetFcMode burn		
		SetLrTarget ACS SUN		
	6.	Verify thrust time set to 16 min.		
	-	SetThrustTime 960		
	7.	Verify thruster enable command matches SunMvrBegin time in OEF.		
		SetThrustEnable ENABLE		
		SunMvrBegin		
	8.	Verify 15 minutes after thrusters enabled: thrusters disabled, cat bed heaters off, thrust time set		
		to 0, FC mode Mission.		
		SetThrustEnable DISABLE		
		SetHTRCmd CATBED_5N_HTR,OFF		
		SetThrustTime 0		
		SetFcMode Mission		
Battery Cell	1	There will be battery cell balancing every 2 out of 3 orbits. Battery cell balancing this orbit? Y/N	09/28/2016	CLR
Balancing	1.		03/20/2010	OLK
Dalaticity	2.	Verify charging cycle (Long eclipse flag=TRUE) is 90 minutes long.		
	3.	Verify the first command sets the long eclipse flag to TRUE, the second command sets the flag to		
		FALSE.		
	4.	Verify P/L is in HVSTANDBY or HVENG.		
	5.	Verify no charging cycle within 2 hours of ASCENDING or DESCENDING macro execution.		
	6.	Verify no charging cycle within 1 hour of maneuver.		
	7.	Verify no charging cycle during an eclipse.		
		, , ,		

## IBEX Command Approval Checklist C. Reno

Rev 16b Last update 06/08/2016

Cmd Violation	1.	Review CCVR. If you have any questions Reply All to the ATS Approval email and ask the team.	09/28/2016	CLR
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Activity	Anomaly Response : Non-nominal burn	Date Completed	Completed By
•	If the maneuver has not occurred or the spacecraft pointing as designated by either the star tracker or coarse Sun sensor is off by more than 2 degrees from the expected pointing, an anomaly has occurred.  • If the spacecraft is in Contingency state all stored commands are flushed from the command queue. Follow standard anomaly process.  • If the Sun maneuver did not occur and the spacecraft is in either Science or Housekeeping state follow the steps below.  • If a partial Sun maneuver has occurred and the spacecraft is in either Science or		
•	Housekeeping state follow the steps below.  Please note that there is no anticipated hardware damage associated with exceeding the 12.5 degree constraint; this constraint is in place because we should not exceed the largest pointing achieved thus far in the mission.		
1.	If the payload is in HVSTANDBY, LVENG or OFF, and there are no commands loaded to bring it to HVSCI voltages, no operational pointing constraints will be violated. Follow		
2.	standard anomaly process.  If the payload in HVSCI mode or there are uploaded commands to bring the payload to HVSCI mode,  a. Determine current off Sun pointing. If the off Sun angle has already exceeded 12.5 degrees, the MOC should notify the MOM and immediately send the DESCENDING command suite in real-time, as described below. If this cannot be done in the contact where the pointing anomaly was discovered, another contact will be planned as soon as possible to execute these commands.  @CEU_MACRO_EXEC DESCENDING_PL1 (<1 min)  @CEU_MACRO_EXEC DESCENDING_HI (~8 min)  @CEU_MACRO_EXEC DESCENDING_HI (~8 min)  @CEU_MACRO_EXEC DESCENDING_HI (~8 min)  Please note: The only scenario where hitting 12.5 degrees is expected is when a perigee Sun precession maneuver is completely missed after an inertial apogee maneuver which occurs late in the maneuver window (near apogee + 10 hours).  b. If current Sun pointing is below 12.5 degrees, the ISOC should input ST Sunpointing angle into ibex_rotate to determine the Sun angle at the time of DESCENDING.  Sun-Angle at payload DESCENDING_  c. If 12.5 degrees is not exceeded by the time of DESCENDING, no payload-specific action is needed for this arc. Follow standard anomaly response process.  d. If 12.5 degrees is exceeded by the time of DESCENDING, a new command set should be sent.  i. The ISOC will generate the new STF which has the DESCENDING commands executing early such that 12.5 degrees is not exceeded while the payload is in HVSCI.  ii. The MOC will create an associated ATS.  iii.Approval is needed by the MOM, MOC & ISOC prior to upload.  iv. The MOC will assess whether an additional pass is needed in order to upload the commands prior to exceeding the 12.5 degree constraint. The onboard DESCENDING commands do not need to be deleted.		